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IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

Applicants: Mannava, et al.

Serial No: 08/719,341

Filed: September 25, 1996

For: LASER SHOCK PEENED GAS TURBINE

ENGINE COMPRESSOR AIRFOIL EDGES)

Hon. Assistant Commissioner for Patents Washington, D.C. 20231

SIR:

AUG 1 0 1998

Group Art Unit:

Verdier Examiner: Chillicke robbins

> I hereby certify that this correspon specified with the United States Postal Service with sufficient

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of person signing certificate

AMENDMENT

In response to the Office Action mailed June 8, 1998, please make the below-identified amendments and consider the following remarks: RECEIVED

IN THE CLAIMS:

AUG 1 3 1998

Please amend the following Claims as indicated. April 2200

(TWICE AMENDED) A compressor blade as claimed in claim 16 further comprising:

a third laser shock peened surface located opposite said first laser shock peened surface such that said first and third laser shock peened surfaces are located along pressure and suction sides of said leading edge respectively,

a third region having deep compressive residual stresses imparted by laser shock peening (LSP) extending into said airfoil from said third laser shock peened surface,

a fourth laser shock peened surface located opposite said second laser shock peened surface such that said second and fourth laser shock peened surfaces are located along pressure and suction sides of said trailing edge respectively, and

said third and fourth regions having deep compressive residual stresses imparted by laser shock peening (LSP) extending into said airfoil from said third and fourth laser shock peened surfaces respectively.

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